

Referee #1

- 1) Query: It would be better to include a Figure, or to modify Figure 1, to show the position of the loudspeakers. If I have well understood there are four couple of loudspeakers, each couple emit sound in phase, and the different couples are arranged like in a windrose at 45, however, it would be more immediate with an image.

Response: the following figure has been added the manuscript.



(a)

(b)

Figure 1. Figure (a): Photograph of the acoustic excitation chamber with eight loud- speakers that are placed in flatwise manner to generate the necessary acoustic pressure in order to perturb the jet shear layer at the nozzle exit.

- 2) Query: Regarding the images some more details should be given like the resolution and the acquisition frequency so that the reader could understand which time span is associate to the 1000 images pairs and, consequently, what is the time span of the average images reported in the paper.

Response: The following text has been added to the manuscript:

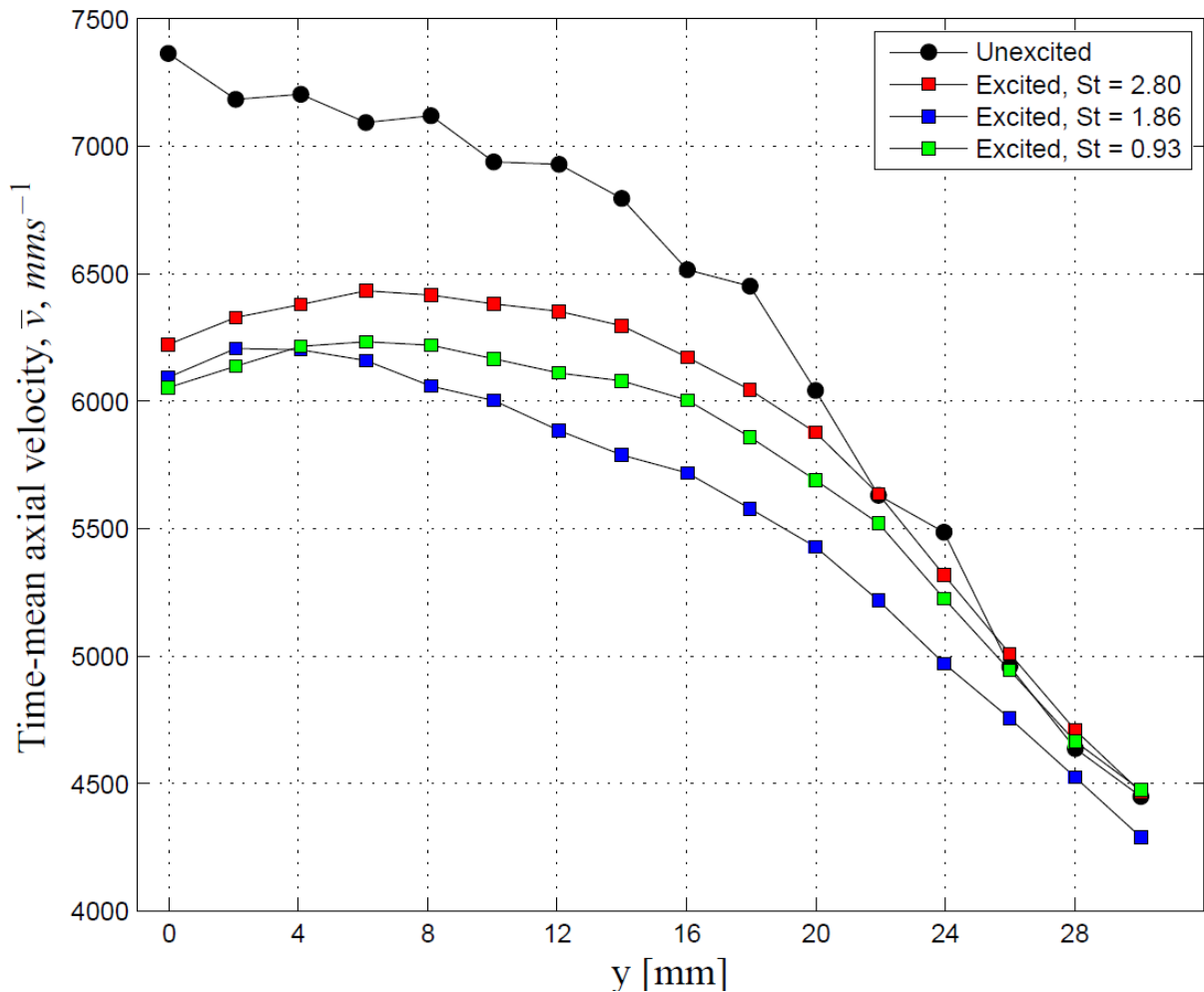
“The PIV system was limited to the frequency of cameras, which was 4Hz. Therefore 1000 image pairs were recorded at a frequency of 4Hz over a period of approximately 250s.”

- 3) Query: Regarding the velocity profiles shown in Figure 4, it is not clear at what distance y are calculated or if they are an average along the y direction. Please explain. Additionally, would the feature shown here (larger velocities for the unexcited jet) be valid also at other distance y or this is a y -dependent feature?

Response: The horizontal axis in Fig. 4 (Fig. 5 in revised manuscript) was mislabelled “ x ” instead of “ y ”. This has now been corrected. The following text has been added to the manuscript

“For these measurements, the light sheets were placed parallel to the y -axis, so that the jet centreline is measurable. The results show the average axial

velocities along the jet centerline axis for the case of an unexcited jet and for excited jets at different St numbers.”



4) Query: Authors should give an estimation of the uncertainties associated to the vorticity, helicity and to the helicity angle. This is an important information useful in the discussion (page 1416 lines 4-8) in which an angle of 178.5 is considered different from 180. Is this coherent with the uncertainty estimates?

Response: The following text was added to the manuscript

“The relative uncertainties in the vorticity components w_x , w_y and w_z were 19%, 16% and 6% respectively. For helicity, the relative uncertainty was found to be approximately 13%.”

The text has also been revised to state that within the limits of accuracy of measurements, 178.5 degrees is not statistically different from 180 degrees.

5) Query: What particle tracers are used as seeding

Response: We have added the following sentence to the manuscript:

“Water droplets generated by ultrasonic humidifier was used as tracer particles.”

Referee 2

Query: There is a definite second peak of 90 degrees in fig 7(b)-(d). Could the authors comment on this?

Response: As far as we know there is no known theory for a preference in helicity angle other than 0, 90 or 180 degrees. No change has been made to the manuscript.

Query: Do the authors have some comments on whether the distribution of helicity angles could be controlled through acoustic excitations?

Response:

Clearly this is a very insightful speculation inspired by the implications of the shape of the distributions in Fig 7 (b) to (d). The hypothesis that acoustic excitation can be used to control helicity is partially supported by these results, and also from our motivation in using the acoustic excitation to form resonant helical structures. The suggested dynamical responses outside the Strohal regime studied would require another experimental campaign. We might possibly undertake this in the future if we revisit this configuration. We would like to thank the referee for the insight.

No changes have been made to the manuscript since this would require further investigations that are beyond the scope of the current study.

Referee 3

Query: An error on the labelling on the axis in Fig 4?

Response: The horizontal axis should have been labelled “y”, not “x”. This has now been corrected in the manuscript.

Dual plane PIV investigation of acoustically excited jets in a swirl nozzle

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Abstract. A novel dual-plane dye laser PIV technique used to analyze helicity and energy dissipation in an unexcited turbulent swirling jet of pressurized cold air has established that regions within the flow field of the jet exhibiting high helicity are correlated regions of high turbulent kinetic energy dissipation. This PIV configuration provides estimates of all components of the velocity gradient tensor, facilitating calculation of the helicity from the vorticity components. Application of this novel dual-plane PIV technique is extended in this study to investigate helical structures in a turbulent swirling jet where the underlying shear flow is subjected to external acoustic sinusoidal forcing in a plane perpendicular to the central axis of the jet. It was found that acoustic excitation had a significant effect on the mean velocity profile parallel to the direction of the jet. The horizontal forcing resulted in the generation of vorticity that was skewed with a pitch that favored a distribution of angles around 90° with respect to the velocity vector. The distribution of the time-averaged helicity angle indicated organized helical activity, but such activity is not dominated by large-scale coherent structures of maximal helicity.

1 Introduction

Helical structures are a common phenomenon in geophysical flows, for example, in magnetohydrodynamics (Moffatt, 1978), storm systems (Lilly, 1986), underwater gravity currents (Sumner et al., 2014) and atmospheric inertia-gravity waves (Viúdez and Dritschel, 2006). It is well-established that helical coherent structures play a crucial role in turbulent flows. However, the role of helicity in three-dimensional turbulent flows is still enigmatic. Complex interactions can occur between different scales of the motion. In addition, nonlinearity and intermittency influence the scaling-laws.

One of the main effects of helicity in neutral isotropic turbulence is the inhibition of energy transfer at large eddy sizes. Kraichnan (1973) investigated analytically the interaction of two helical waves, concluding that the presence of a non-zero mean helicity in a turbulent flow inhibits the energy transfer to small scales. This was confirmed numerically by André and Lesieur (1977). Thus
25 homogeneous turbulence with significant mean helicity exhibits a much slower rate of decay of turbulence and its corresponding energy than its zero-mean helicity counterpart.

When rotational effects are present significant differences exist between helical and non-helical turbulent flows (Pouquet and Mininni, 2010). Orlandi (1997) carried out direct numerical investigations of helicity fluctuations and the production of turbulent energy in rotating and non-rotating
30 pipes. Results showed that helical dynamics were important across all scales and that there existed a correlation between maximal helicity and low energy dissipation. Such an occurrence is an attractive prospect as it implies that effective dispersion of heat or mass using low mechanical energy consumption is possible with the use of helical flows with maximal helicity, that is, via helical coherent structures. In contrast, three-dimensional laminar studies of flow through a staggered herringbone
35 microchannel at low Reynolds number conducted by Hassell and Zimmerman (2006) revealed an increase in helicity in flows corresponded to an increase in the energy dissipation.

Khomenko et al. (1991) showed that for the case of compressible turbulence large scale helical structures can be generated from small scale turbulent helices. Subsequently, Khomenko and Zimmerman (1994) demonstrated the existence of an inverse cascade of helical structure formation, where large
40 helical structures are formed by small helical structures. Exploitation of this inverse cascade can lead to improved mixing efficiencies due to the dispersion effects of helical coherent structures. In turbulent flows, Zimmerman (1996a) demonstrated that intensified mixing could be achieved with helical coherent structures. Simulations of mixing showed up to a twenty-fold improvement in eddy diffusivities compared to that possible from an equivalent input of energy to base state of isotropic
45 turbulence Zimmerman (1996b). This was attributed to the three-dimensional dispersive property of helical flows, which enhances molecular diffusion due to the creation of strong gradients.

It is well known that two types of instabilities may be exhibited in the shear layer of an unexcited, free jet: the column instability, which causes the whole jet to oscillate as if it were an oscillating elastic column, and the shear layer instability which induces unstable waves in the mixing layer that
50 surrounds the jet core immediately downstream of the nozzle exit. However, as the shear layer instability progresses downstream, it grows and merges, eventually becoming difficult to distinguish from the column instability. The distinguishing feature between these two types of instabilities is their dependence on the resonant frequency at the nozzle exit Reynolds number, Re . This is determined by the non-dimensional Strouhal number, St , given by $St = \frac{fD}{U_e}$, where f denotes frequency,
55 D is the nozzle exit diameter and U_e is the velocity at the nozzle exit. It has been found that St lies between 0.2 to 0.5 for column instability whilst for shear layer instability the proportionality factor of $\frac{St}{\sqrt{Re}}$ is 0.012 to 0.017 (Crow and Champagne, 1971; Drobniak and Klajny, 2002). The degree of

swirl in a flow is characterized by the non-dimensional swirl number, S , given by

$$S = \frac{\int_0^R uwr^2 dr}{R \int_0^R u^2 r dr}, \quad (1)$$

60 where R is the radius of the nozzle, u and w denote the axial and tangential velocity components, and r is the radial coordinate (Sheen et al., 1996).

Regunath et al. (2008) reported on the use of a novel dual-plane dye PIV laser system for investigating energy dissipation and helicity in a turbulent swirling flow. The PIV system comprised a single frequency Nd:YAG laser which generated two parallel light sheet planes of different wavelengths.

65 This was operated in conjunction with three CCD cameras. Results from this study showed that in an unforced turbulent swirling jet high rates of turbulent kinetic energy dissipation correlated with regions exhibiting strong helicity. The purpose of this paper is to extend the work of Regunath et al. (2008) by investigating the role of helicity in a turbulent swirling forced jet. Here, the term "forced jet" implies the use of an external force to perturb the shear layer in order to generate helical waves.
70 By applying sinusoidal perturbations in the form of acoustic waves, helical structures are formed. Further understanding of the interplay between helicity and turbulence in this setting will impact on our knowledge of helicity-turbulence interactions in a range of geophysical situations.

The experimental setup, together with a brief overview of the methodology of the particle image velocimetry (PIV) technique is given in Section 2. Results from the dual-plane PIV analysis of
75 acoustically excited jets are presented in Section 3 and conclusions are summarized in Section 4.

2 Experimental setup and statistical analysis of PIV data

PIV enables the instantaneous velocity field of a flow to be recorded by means of a technique based upon cross-correlations. Originally used to obtain two-dimensional velocity fields, developments over the last decade have led to the successful realization of three-dimensional velocity fields
80 (Lynch and Thurow, 2011). Statistical properties of the mean and turbulent components can be estimated from ensemble averaging (Ullum et al., 1998). However, accuracy of the statistical analysis depends on the sample size over which the ensemble averages are formed (Uzol and Camci, 2001). Depending on the degree of variability of the flow, the number of image pairs required to provide the desired level of statistical accuracy typically ranges from 200 to 750. A stereoscopic, i.e. three-
85 dimensional, PIV technique used in conjunction with a dual-plane dye laser system can be used to estimate all nine velocity gradient components, $\frac{\partial u_i}{\partial x_j}$, thus facilitating the determination of all three vorticity components, ω_x , ω_y and ω_z . The experimental configuration used is described in detail in Regunath et al. (2008). In the current study the experimental apparatus is enhanced by a system of eight loudspeakers deployed so as to induce sinusoidal perturbations in the form of acoustic waves
90 [\(Fig. 1\)](#). These sinusoidal perturbations give rise to a forced jet.

Figure 1: Photographic image of the acoustic excitation chamber with eight loudspeakers that are placed in flatwise manner to generate the necessary acoustic pressure in order to perturb the jet shear layer at the nozzle exit.

Figure 2: Schematic of the experimental setup. The swirl nozzle is located within the wooden chamber.

2.1 Experimental design

Full details of the basic experimental configuration are given in Regunath et al. (2008), so only a brief outline will be given here. A centrifugal compressor supplied the working fluid of pressurized cold air through a twisted vane swirl nozzle to impart both a tangential and an axial momentum within the flow field, thus giving rise to a swirling motion (Fig. 1-2). Water droplets generated by an ultrasonic humidifier were used as tracer particles. The exit diameter, D , of the nozzle had a width of 32 mm. Dual-plane dye laser PIV measurements were made in the near-jet region, $0 < y/D < 1$, where y denotes the streamwise direction. Vertical light sheets in two planes slightly displaced from one another were used to illuminate the flow field. The center of the green light sheet plane, $\lambda = 532$ nm, was aligned at the centerline of the jet ($z = 0$ mm), whilst the center of the yellow light sheet plane, $\lambda = 572$ nm, was positioned along the plane $z = -1$ mm (Fig. 23). The thickness of each light sheet was measured to be 2 mm, thus the yellow light sheet extended from $z = -2$ mm to $z = 0$ mm and the green light sheet extended from $z = -1$ mm to $z = 1$ mm. Although the light sheets overlapped, the optical filters that were placed in front of the cameras only permitted one of the two wavelengths, $\lambda = 532$ nm or $\lambda = 572$ nm, to be captured. The dual-plane dye laser PIV system was used to obtain stereoscopic measurements on the green light sheet. Planar two-dimensional measurements were made on the yellow light sheet. Ensemble time averages based on 1000 PIV image pairs of the instantaneous velocity vectors were formed to produce the mean velocity field within each measured plane. The PIV system was limited to the frequency of cameras, which was 4Hz. Therefore 1000 image pairs were recorded at a frequency of 4Hz over a period of approximately 250s.

In order to generate the acoustic perturbations required for the present study an acoustic excitation chamber comprising eight loudspeakers was arranged circumferentially around the jet nozzle exit. Helical acoustic waves were generated via inputted sinusoidal signals, of frequency f , that were

Figure 3: Schematic of the swirling jet chamber showing the orientation and position of the two light sheets.

phase shifted by $\frac{\pi}{2}$ between each adjacent speaker resulting in a sequential pressure variation at
115 the nozzle exit. A four-channel high-performance power amplifier with a maximum power of $4 \times$
120 120 W was used to amplify the generated sinusoidal signals. A trigger signal connected to the PIV
synchronizer enabled the PIV system to capture images at appropriate time intervals. Measurements
were made at frequencies of 200 Hz, 400 Hz and 600 Hz to investigate the effectiveness of input
frequency for generating helical structures in the resultant flow field. These frequencies correspond
to Strouhal numbers, St , of 0.93, 1.86 and 2.80 respectively. A control experiment without acous-
tic excitation was performed for comparison purposes. The measured Reynolds number and Swirl
number for these experimental investigations were kept constant at $Re = 12890$ and $S = 0.43$.

3 Results

The dual-plane dye laser PIV technique was used to illuminate the vertical planes aligned along
125 $z = -1$ mm and $z = 0$ mm which permitted measurements of the velocity vector field to be made.
The single camera vector field constructed from measurements obtained from the plane $z = -1$ was
used in conjunction with the stereoscopic vector field from the $z = 0$ mm plane to compute all 9
velocity gradients in the $z = 0$ mm plane. The velocity gradient fields were obtained from second-
order central differences of the measured velocity component values within each plane and from
130 first-order forward differences between the two illuminated planes aligned parallel to the y -axis
(Fig. 23).

Contours of the magnitude of the mean velocity field on the plane $z = 0$ mm for an unforced
jet and for forced jets for a range of Strouhal numbers ($St = 0.93, 1.86$ and 2.80) are presented in
Fig. 34. To facilitate comparisons, uniform color scaling has been used in each sub-figure. The jet
135 structure in each case is qualitatively similar, although a larger degree of irregularity is noticed in
the instance of the unexcited jet and the magnitude of the velocity field at the nozzle exit is greater.

Figure 4: Magnitude of velocity field on $z = 0$ mm plane: (a) for unexcited jet; (b) for excited jet
with $St = 0.93$; (c) for excited jet with $St = 1.86$; and, (d) for excited jet with $St = 2.80$.

To further investigate the effect of excitation on the swirling jet, the decay of the time-mean
axial velocity along the jet centerline is compared for excited and unexcited cases in Fig. 45. For
these measurements, the light sheets were placed parallel to the y -axis, so that the jet centreline is
measurable. The results show the average axial velocities along the jet centerline axis for the case
140 of an unexcited jet and for excited jets at different St numbers. The velocities of the excited jets
at the nozzle exit were found to be 15% to 17% lower than the exit velocity of the unexcited jet.
After a slight increase in mean axial velocity in the vicinity of the nozzle exit, the axial velocities
of the excited jets then decreased monotonically. For example, the axial velocities of the excited jets

145 at $St = 2.80$ ($f = 600$ Hz) and $St = 0.93$ ($f = 200$ Hz) show that the mean axial velocity increases
 across the first 6 mm downstream of the exit nozzle. The axial velocity of the excited jet with $St =$
 1.86 ($f = 400$ Hz) increases over the first 2 mm then decreases along the jet centerline. We can
 attribute this behavior of the velocity profiles as response to the perturbations applied to the jet at the
 nozzle exit since for the case of the unexcited jet the maximal velocity was found to occur at the exit
 150 itself, rather than at a downstream location.

Figure 5: Decay of time-mean axial velocities along the jet centerline axis for the case of an unexcited
 jet and for excited jets at different St numbers.

The dual-plane dye laser PIV technique permits us to compute all the velocity and vorticity com-
 ponents over the $z = 0$ plane, thus we can calculate the helicity density, h , which is defined as the
 scalar product of the local velocity vector, \mathbf{u} , and the local vorticity vector, $\boldsymbol{\omega}$, i.e. $h = \mathbf{u} \cdot \boldsymbol{\omega}$. [The](#)
[relative uncertainties in the vorticity components \$\omega_x, \omega_y\$ and \$\omega_z\$ were 19%, 16% and 6% respectively.](#)
 155 [For helicity, the relative uncertainty was found to be approximately 13%.](#) We are also able to deter-
 mine the helicity angle, θ from

$$\cos \theta = \frac{\mathbf{u} \cdot \boldsymbol{\omega}}{\|\mathbf{u}\| \|\boldsymbol{\omega}\|}. \quad (2)$$

Fig. 5-6 shows the computed mean helicity density, \bar{h} , for the unexcited and excited jets respec-
 tively. It can be seen that the helicity density values at the nozzle exit have predominantly negative
 160 values, which, since helicity density is a pseudoscalar quantity, is indicative of clockwise rotation
 (left-hand chirality). The color scales for these figures represent the helicity density and have been
 plotted on a uniform scale of $-3 \times 10^7 \text{ mms}^{-2}$ to $2 \times 10^6 \text{ mms}^{-2}$. It is observed that the largest
 magnitude of the average helicity density is associated with the unexcited jet at the nozzle exit.

Figure 6: Average helicity density, \bar{h} , on $z = 0$ mm plane: (a) for unexcited jet; (b) for excited jet
 with $St = 0.93$; (c) for excited jet with $St = 1.86$; and, (d) for excited jet with $St = 2.80$.

Fig. 6-7 shows plots of average helicity angle, $\bar{\theta}$, for the unexcited and excited jets across the $z = 0$
 165 plane. Plots of $\bar{\theta}$ provide insight into the occurrence of helical coherent structures (HCS). HCS are
 structures that have maximal helicity. These are defined as those where the angle between velocity,
 \mathbf{u} , and vorticity, $\boldsymbol{\omega}$ is either 0° (indicated by blue) or 180° (indicated by red).

Fig. 6-7 show the occurrence of a few regions where the helicity angle is close to 0° (colored
 dark blue), and some (colored dark red) that indicate a helicity angle around 180° . A bar chart
 170 logging the occurrence of the maximal helicity angle for the $z = 0$ plane is presented in Fig. 78. The
 histograms of the unexcited and excited jets all show that there are no counts of a helicity angle of 0° .

Figure 7: Average helicity angle, $\bar{\theta}$, on $z = 0$ mm plane: (a) for unexcited jet; (b) for excited jet with $St = 0.93$; (c) for excited jet with $St = 1.86$; and, (d) for excited jet with $St = 2.80$.

However, for the excited jet at $St = 1.86$ an occurrence of a helicity angle in the range $180^\circ \pm 5^\circ$ is logged. Upon further inspection it was found that this angle was in fact 178.5° , ~~thus confirming that there were no occurrences of~~ which is not significantly different from 180° due to limitations on accuracy of measurements. In fact, the modal helicity angle was found to be around 90° , ~~which shows that there is no evidence to support the existence of large-scale helical structures with maximal helicity.~~

Figure 8: Bar charts of the average helicity angle, $\bar{\theta}$, in degrees, on the $z = 0$ mm plane: (a) for unexcited jet; (b) for excited jet with $St = 0.93$; (c) for excited jet with $St = 1.86$; and, (d) for excited jet with $St = 2.80$.

4 Conclusions

Preliminary experimental results of acoustic excitation on swirling jets have been reported. A twisted vane swirling jet was used to generate the swirling flow fields. An acoustic excitation method comprising eight loudspeakers that operated in pairs was used to generate sinusoidal signals to excite the swirling jets by perturbing the shear layer. To measure the effect of acoustic excitation on the helical properties of the resultant flow field, the novel dual-plane dye laser PIV method was used. This technique was used to illuminate the $z = -1$ mm and $z = 0$ mm vertical planes to measure the fluid velocities. Three different experiments were carried out at frequencies of 200 Hz, 400 Hz and 600 Hz, corresponding to Strouhal numbers of $St = 0.93$, $St = 1.86$ and $St = 2.80$. A further experiment without the effect of acoustic excitation was also conducted to allow comparison of results for excited and unexcited jets. By observing the plots of velocity magnitude, helicity density and helicity angle, it was found that the jets under acoustic excitation all had different velocity profiles when compared to the jet without any acoustic excitation. Furthermore, the distributions of average helicity angle, $\bar{\theta}$, revealed a greater degree of skewness between the velocity and vorticity vectors than for the unforced case. Evidence for the existence of measurable organized helical structures was found, though the helical vortices were not dominated by coherent structures exhibiting maximal helicity in either unexcited nor excited swirling jets ~~since no occurrences of 0° or 180° helical angles were found.~~ However, some coherent structures with close to maximal helicity were found to occur.

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